

BEJ

DATE		TAIL NUMBER		STATION		AIRLINE CARD NO.		BOEING CARD NO. 49-010-00-00	
SKILL ENGIN	WORK AREA APU COMPARTMENT	RELATED TASK		VERSION 1.1	THRESHOLD 2 YRS	REPEAT 2 YRS		PHASE	
TASK GENERAL VISUAL		TITLE APU MOUNTS				APPLICABILITY AIRPLANE ENGINE ALL ALL			
ZONES 315 316		ACCESS 315A							
SYSTEMS MPD ITEM: 49-010-00 PERFORM A GENERAL VISUAL INSPECTION OF THE APU MOUNTS FOR GENERAL CONDITION AND SECURITY OF INSTALLATION. A. References (1) AMM TASK 49-13-11-200-801 p601, APU Mounts Inspection (APU Removed)								MECH	INSP
								CUSTOMER FLEET EFFECTIVITY ALL	

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-010-00-00		
<p>TASK 49-13-11-200-803</p> <p>1. <u>APU Mounts Inspection (APU Installed)</u></p> <p>A. Prepare for the Inspection</p> <ul style="list-style-type: none"> (1) Make sure the APU master switch on the P5 forward overhead panel is OFF and install a DO-NOT-OPERATE tag. (2) Open these circuit breakers and attach DO-NOT-CLOSE tags: <ul style="list-style-type: none"> (a) Circuit Breaker Panel, P6-4: <ul style="list-style-type: none"> 1) 6A14 AUX POWER UNIT CONT (b) Circuit Breaker Panel, P6-2: <ul style="list-style-type: none"> 1) 6B19 APU FIRE SW POWER (3) Open the APU Cowl Door, 315A: <ul style="list-style-type: none"> (a) Open the three latches. (b) Open the APU Cowl Door, 315A. (c) Remove the retainer pin from the rod end of the forward hold-open rod on the APU Cowl Door, 315A. (d) Remove the retainer pin from the spring clip on the aft hold-open rod. (e) Disconnect the two hold-open rods from the two spring clips. (f) Connect the two rod ends of the two hold-open rods to the two brackets in the APU compartment. (g) Install the two retainer pins in the two rod ends. <p>B. Procedure</p> <ul style="list-style-type: none"> (1) Do these steps to inspect the APU mounts: <ul style="list-style-type: none"> (a) Make sure all connections for the APU mounts are tight. (b) Visually examine these parts of the APU mounts that you can get access for corrosion, cracks and damage: <ul style="list-style-type: none"> 1) Strut assemblies (APU mounts) 2) Vibration isolators 3) Cone bolts and nuts for the vibration isolator. (c) If you find corrosion, cracks or damage to these parts, then you must inspect the APU mounts with the APU removed. To inspect them, do this task: APU Mounts Inspection (APU Removed) (AMM TASK 49-13-11-200-801 p601). <p>C. Put the Airplane Back to Its Usual Condition</p> <ul style="list-style-type: none"> (1) Close the APU Cowl Door, 315A: <ul style="list-style-type: none"> (a) Remove the two retainer pins from the two hold-open rods in the APU compartment. (b) Disconnect the two hold-open rods from the two brackets. (c) Put the two hold-open rods in the two spring clips on the APU Cowl Door, 315A. (d) Install the retainer pin in the rod end of the forward hold-open rod. (e) Install the retainer pin to the spring clip on the aft hold-open rod. (f) Close the APU Cowl Door, 315A. (g) Close the three latches. 					MECH	INSP
CUSTOMER FLEET EFFECTIVITY ALL			SOURCE MRB	<p>APU MOUNTS</p> <p>49-010-00-00</p> <p>PAGE 2 of 3 Oct 10/02</p>		

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-010-00-00	
<p>(2) Remove the D0-NOT-CLOSE tags and close these circuit breakers:</p> <ul style="list-style-type: none">(a) Circuit Breaker Panel, P6-4:<ul style="list-style-type: none">1) 6A14 AUX POWER UNIT CONT(b) Circuit Breaker Panel, P6-2:<ul style="list-style-type: none">1) 6B19 APU FIRE SW POWER <p>(3) Remove the D0-NOT-OPERATE tag from the APU master switch on the P5 forward overhead panel.</p>				MECH	INSP
				CUSTOMER FLEET EFFECTIVITY ALL	

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DATE		TAIL NUMBER	STATION	AIRLINE CARD NO.		BOEING CARD NO. 49-020-00-00		
SKILL ENGIN	WORK AREA APU COMPARTMENT	RELATED TASK		VERSION 1.1	THRESHOLD 8 YRS	REPEAT 8 YRS	PHASE	
TASK INSPECTION		TITLE APU MOUNTS			APPLICABILITY AIRPLANE ENGINE ALL ALL			
ZONES 315 316		ACCESS 315A						
<p>SYSTEMS</p> <p>MPD ITEM: 49-020-00 PERFORM A DETAILED INSPECTION OF THE APU MOUNTS.</p> <p>A. References</p> <ul style="list-style-type: none"> (1) AMM TASK 49-11-00-000-801 p401, APU Power Plant Removal (2) AMM TASK 49-11-00-400-801 p401, APU Power Plant Installation (3) AMM TASK 49-13-11-000-802 p401, APU Mounts Removal (4) AMM TASK 49-13-11-400-802 p401, APU Mounts Installation 							MECH	INSP
CUSTOMER FLEET EFFECTIVITY ALL				SOURCE MRB	APU MOUNTS 49-020-00-00			

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-020-00-00
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TASK 49-13-11-200-801

1. APU Mounts Inspection (APU Removed) (Fig. 601)

A. Procedure

- (1) Do these steps to inspect the APU mounts:
 - (a) Remove the APU. To remove it, do this task: APU Power Plant Removal (AMM TASK 49-11-00-000-801 p401).
 - (b) Visually examine these parts for corrosion, cracks and damage:
 - 1) Strut assemblies (APU mounts)
 - 2) Vibration isolators
 - 3) Cone bolts and nuts for the vibration isolator.
 - (c) If you find corrosion, cracks or damage, then do these steps:
 - 1) Remove the parts for the APU mounts that you find with corrosion, cracks or damage. To remove them, do this task: APU Mounts Removal (AMM TASK 49-13-11-000-802 p401).

NOTE: It is necessary to remove the firewall covers to get access to the APU mounts, support brackets and mounting parts.

- 2) Visually examine the bolts and bushings for corrosion, wear and damage.
 - a) Replace the bolts and bushings that you find with corrosion or damage.
 - b) Replace all the parts that are more than the permitted wear limits shown in (Table 601).

TABLE 49-13-11-993-803

ITEM NUMBER	PART	DIMENSION INNER DIAMETER (ID)/OUTER DIAMETER (OD)	DESIGN LIMITS		WEAR LIMITS		REPAIR
			DIAMETER		PERMITTED WEAR	MAXIMUM CLEARANCE	
			MINIMUM	MAXIMUM			
			INCH (MM)	INCH (MM)	INCH (MM)	INCH (MM)	
1	ROD END	ID	0.3120 (7.92)	0.3125 (7.94)	0.3175 (8.06)	0.0100 (0.25)	*[1]
2	BOLT	OD	0.3115	0.3120	0.3060	0.0100	*[1]

APU Mount Inspection
Table 601

CUSTOMER FLEET EFFECTIVITY ALL	SOURCE MRB	APU MOUNTS 49-020-00-00	PAGE 2 of 7 Jun 10/00
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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-020-00-00
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TABLE 49-13-11-993-803

ITEM NUMBER	PART	DIMENSION	DESIGN LIMITS		WEAR LIMITS		REPAIR
			DIAMETER		PERMITTED WEAR	MAXIMUM CLEARANCE	
			MINIMUM	MAXIMUM			
			INCH (MM) (7.91)	INCH (MM) (7.92)	INCH (MM) (7.77)	INCH (MM) (0.25)	
3	ROD END	ID	0.2495 (6.34)	0.2500 (6.35)	0.2550 (6.48)	0.0100 (0.25)	*[1]
4	BOLT	OD	0.2490 (6.32)	0.2495 (6.34)	0.2435 (6.18)	0.0100 (0.25)	*[1]
5	ROD END	ID	0.3120 (7.92)	0.3125 (7.94)	0.3175 (8.06)	0.0100 (0.25)	*[1]
6	BOLT	OD	0.3115 (7.91)	0.3120 (7.92)	0.3060 (7.77)	0.0100 (0.25)	*[1]
7	STRUT	ID	0.5625 (14.29)	0.5631 (14.30)	-----	-----	*[2]
8	BUSHING	OD	0.4365 (11.09)	0.4370 (11.10)	0.4315 (10.96)	0.0100 (0.25)	*[1]
9	BUSHING	OD	0.5631 (14.30)	0.5638 (14.32)	-----	-----	*[3]
		ID	0.4400 (11.18)	0.4415 (11.21)	0.4465 (11.34)	0.0100 (0.25)	*[1]
10	BOLT	OD	0.2490 (6.32)	0.2495 (6.34)	0.2435 (6.18)	0.0100 (0.25)	*[1]
11	ROD END	ID	0.2497 (6.34)	0.2502 (6.36)	0.2552 (6.48)	0.0100 (0.25)	*[1]

APU Mount Inspection
Table 601

CUSTOMER FLEET EFFECTIVITY ALL	SOURCE MRB	APU MOUNTS 49-020-00-00	PAGE 3 of 7 Jun 10/00
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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-020-00-00
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TABLE 49-13-11-993-803

ITEM NUMBER	PART	DIMENSION	DESIGN LIMITS		WEAR LIMITS		REPAIR
			DIAMETER		PERMITTED WEAR	MAXIMUM CLEARANCE	
			MINIMUM	MAXIMUM			
			INCH (MM)	INCH (MM)	INCH (MM)	INCH (MM)	
12	STRUT	ID	0.5625 (14.29)	0.5631 (14.30)	-----	-----	*[2]
13	BUSHING	OD	0.4365 (11.09)	0.4370 (11.10)	0.4315 (10.96)	0.0100 (0.25)	*[1]
14	BUSHING	OD	0.5631 (14.30)	0.5638 (14.32)	-----	-----	*[3]
		ID	0.4400 (11.18)	0.4415 (11.21)	0.4465 (11.34)	0.0100 (0.25)	*[1]

- *[1] *[1] REPLACE WHEN WORN
- *[2] *[2] OVERSIZE STRUT HOLE MUST NOT BE MORE THAN 0.625 INCH (15.88 MM) IN DIAMETER
- *[3] *[3] REPLACE WITH OVERSIZE BUSHING

APU Mount Inspection
Table 601

- 3) Examine the four bolts and four lockwashers that attach each housing assembly to each vibration isolator for tightness and missing part(s).
 - a) If it is necessary, tighten the bolts or replace the missing part(s).
- 4) Visually examine the surface of each vibration isolator for scratches, nicks, burrs, corrosion, galling, fretting and wear.
 - a) If the individual damaged area is more than 0.50 inch (12.70 mm) diameter by 0.20 inch (0.51 mm) depth or 1.00 inch (25.40 mm) length by 0.10 inch (2.54 mm) width by 0.020 inch (0.51 mm) depth, replace the vibration isolator.
 - b) If the total damaged area is more than 15% of the total surface area for the vibration isolator, replace the vibration isolator.

CUSTOMER FLEET EFFECTIVITY ALL	SOURCE MRB	APU MOUNTS 49-020-00-00	PAGE 4 of 7 Jun 10/00
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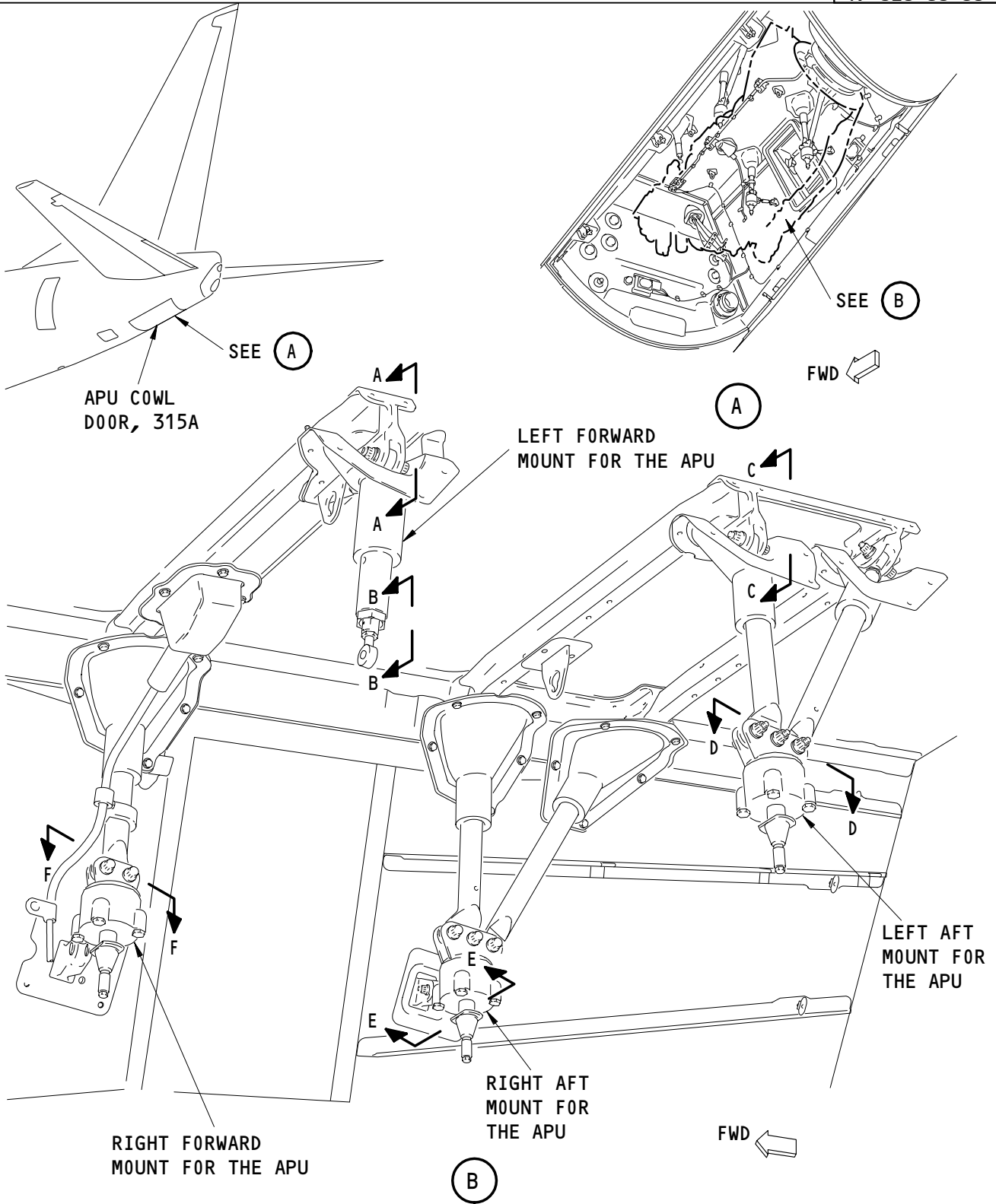
BEJ



DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-020-00-00	
<p>5) Visually examine the threads of the cone bolts and nuts on the vibration isolator for galling, wear and damage. a) Replace the vibration isolator and nuts that you find with galling, wear or damage.</p> <p>6) Install the new or serviceable part(s) for the APU mounts. To install it, do this task: APU Mounts Installation (AMM TASK 49-13-11-400-802 p401).</p> <p>(d) Make sure all connections for the APU mounts and support brackets are tight.</p> <p>(e) Install the APU. To install it, do this task: APU Power Plant Installation (AMM TASK 49-11-00-400-801 p401).</p>				MECH	INSP
				CUSTOMER FLEET EFFECTIVITY ALL	

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AIRLINE CARD NO.
BOEING CARD NO.
49-020-00-00



APU Mounts Inspection
Figure 601 (Sheet 1)

CUSTOMER FLEET EFFECTIVITY

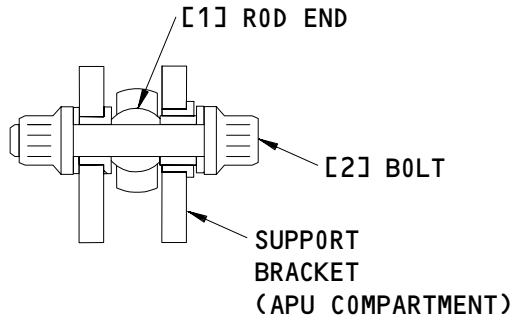
ALL

SOURCE

MRB

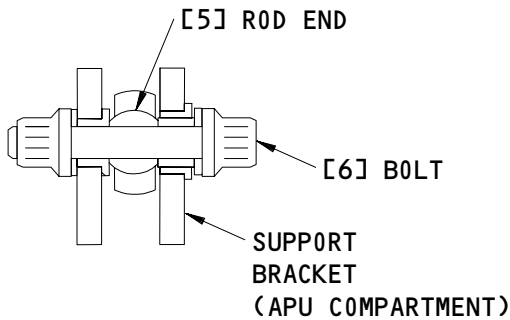
APU MOUNTS

49-020-00-00



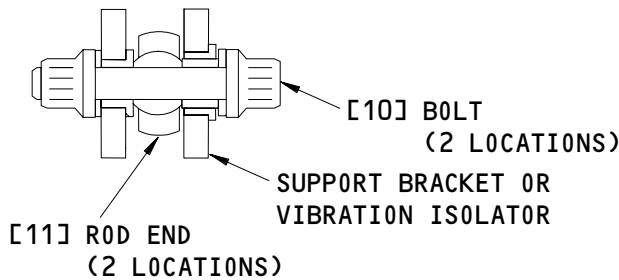
(2 LOCATIONS ON THE TWO FORWARD MOUNTS)

A-A

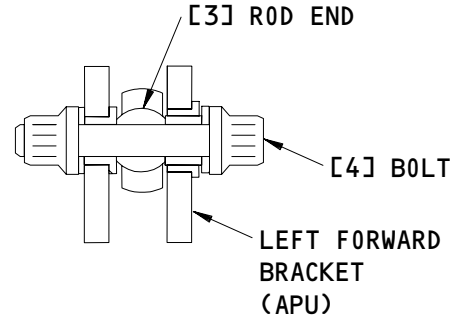


(4 LOCATIONS ON THE TWO AFT MOUNTS)

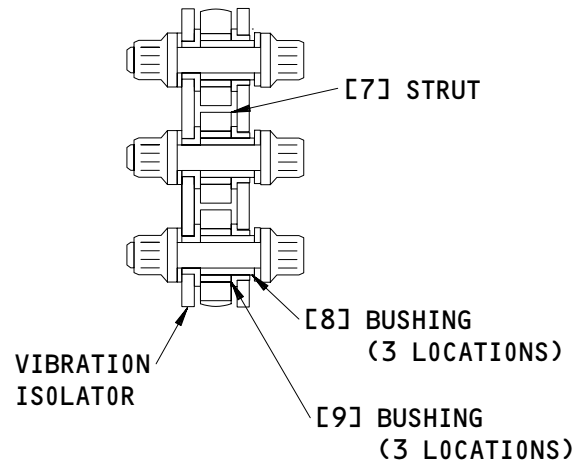
C-C



E-E

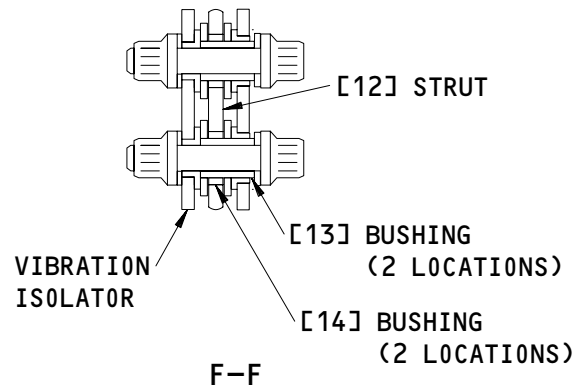


B-B



(2 LOCATIONS ON THE TWO AFT MOUNTS)

D-D



F-F

APU Mounts Inspection
Figure 601 (Sheet 2)

CUSTOMER FLEET EFFECTIVITY

ALL

SOURCE

MRB

APU MOUNTS

49-020-00-00

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DATE		TAIL NUMBER		STATION		AIRLINE CARD NO.		BOEING CARD NO. 49-030-00-00		
SKILL ENGIN	WORK AREA APU COMPARTMENT	RELATED TASK		VERSION 1.1	THRESHOLD APU CNG	REPEAT		PHASE		
TASK INSPECTION		TITLE SIGMA SEAL				APPLICABILITY AIRPLANE ENGINE ALL ALL				
ZONES 316			ACCESS 315A							
<p>SYSTEMS</p> <p>MPD ITEM: 49-030-00 PERFORM A DETAILED INSPECTION OF THE SIGMA SEAL. (AFTER APU REMOVAL).</p> <p>A. References</p> <ul style="list-style-type: none"> (1) AMM TASK 49-11-00-000-801 p401, APU Power Plant Removal (2) AMM TASK 49-11-00-400-801 p401, APU Power Plant Installation (3) AMM TASK 49-15-11-000-801 p401, Air Inlet Seal Removal (4) AMM TASK 49-15-11-400-801 p401, Air Inlet Seal Installation 								MECH	INSP	
								CUSTOMER FLEET EFFECTIVITY ALL		

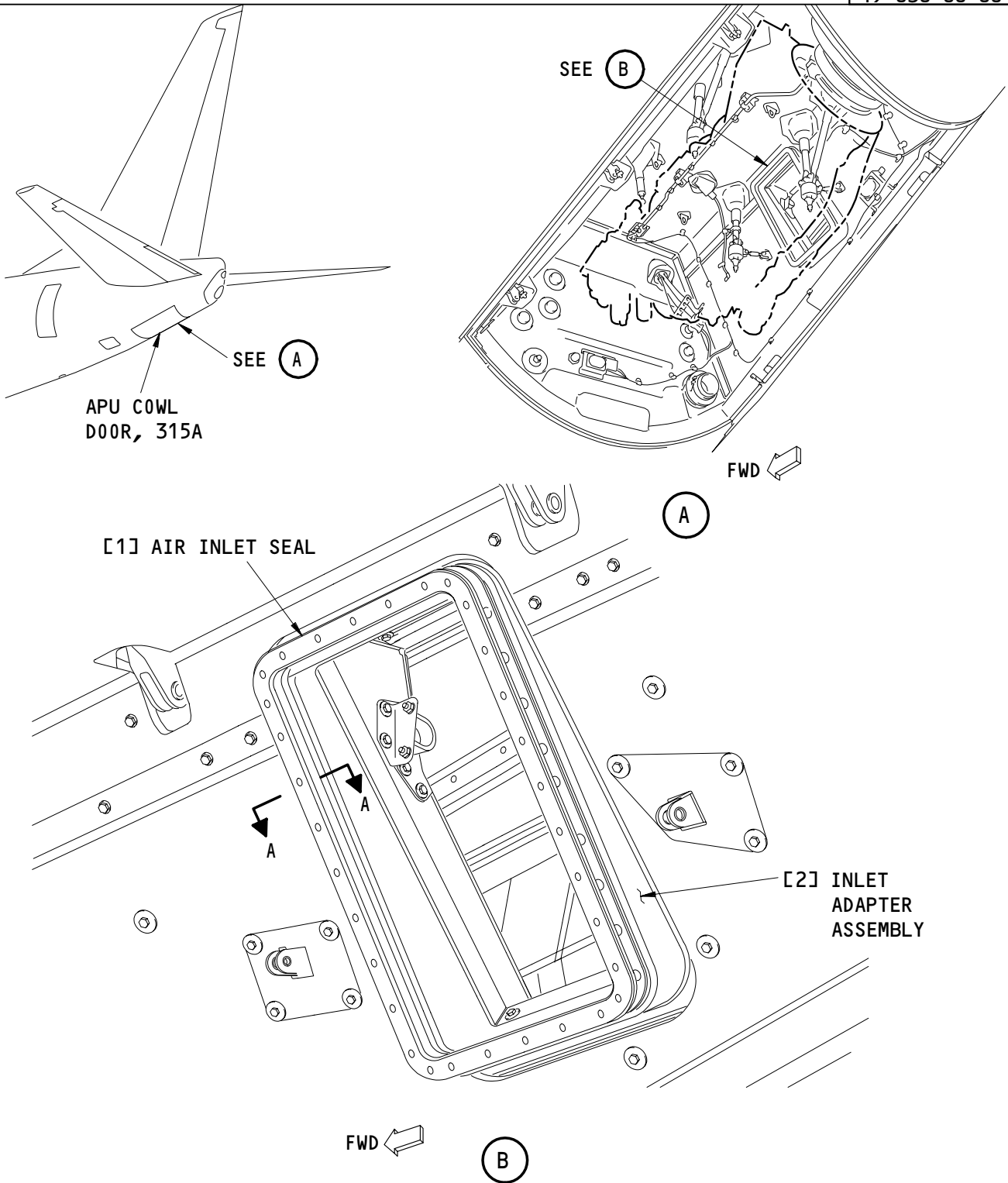
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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-030-00-00	
<p>TASK 49-15-11-200-801</p> <p>1. <u>Air Inlet Seal Inspection</u> (Fig. 601)</p> <p>A. Prepare for the Removal</p> <p>(1) Remove the APU. To remove it, do this task: APU Power Plant Removal (AMM TASK 49-11-00-000-801 p401).</p> <p>B. Procedure</p> <p>(1) Do these steps to inspect the air inlet seal [1]:</p> <p>(a) Examine the air inlet seal [1] for any signs of folding, tears or deformation.</p> <p>(b) Measure the height of the air inlet seal [1].</p> <p><u>NOTE:</u> The height from the bottom to the top of the air inlet seal [1] must be 1.375-1.625 inches (34.9-41.3 mm).</p> <p>(c) Make sure the retainer plate and stiffener plate are attached to the air inlet seal [1].</p> <p><u>NOTE:</u> You can find the retainer plate between the 32 screws and the air inlet seal [1]. You can find the stiffener plate on the bottom of the air inlet seal.</p> <p>(d) Examine the mating surfaces of the air inlet seal [1] for wrinkles, bubbles, unwanted materials or wear damage through the top rubber layer of fiberglass.</p> <p>(e) Examine the seven rubber layers of fiberglass for any separations, missing materials, cracks and tears.</p> <p>(f) If you find any of the above damage or the height of the air inlet seal [1] is not in the limits, replace the air inlet seal. These are the tasks: Air Inlet Seal Removal (AMM TASK 49-15-11-000-801 p401), Air Inlet Seal Installation (AMM TASK 49-15-11-400-801 p401).</p> <p>(2) Install the APU. To install it, do this task: APU Power Plant Installation (AMM TASK 49-11-00-400-801 p401).</p>				MECH	INSP
CUSTOMER FLEET EFFECTIVITY ALL		SOURCE MRB	<p>SIGMA SEAL</p> <p>49-030-00-00</p> <p>PAGE 2 of 4 Jun 10/02</p>		

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AIRLINE CARD NO.
BOEING CARD NO. 49-030-00-00



**Air Inlet Seal Inspection
Figure 601 (Sheet 1)**

CUSTOMER FLEET EFFECTIVITY

ALL

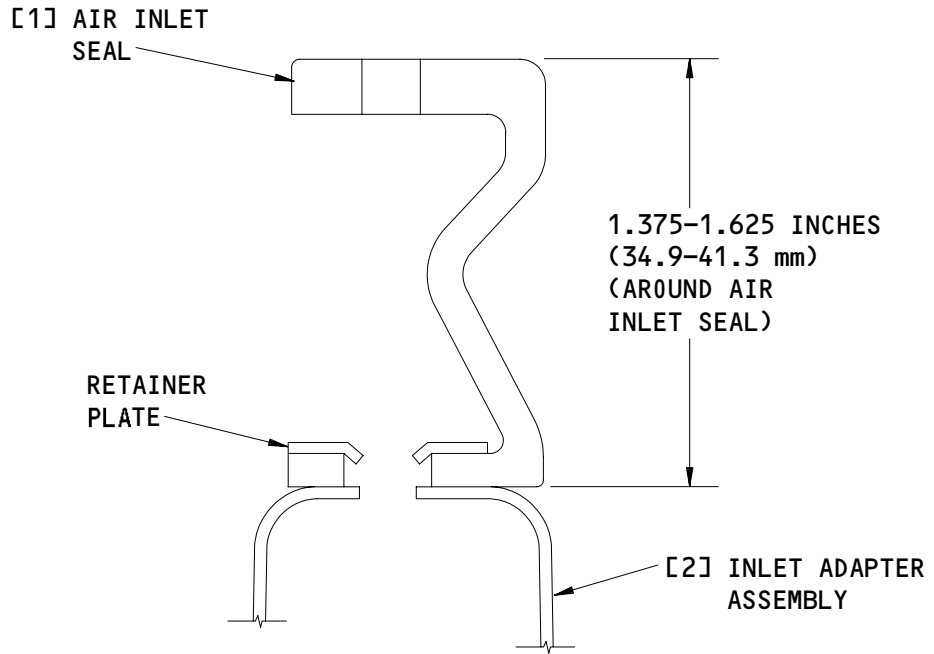
SOURCE

MRB

SIGMA SEAL

49-030-00-00

M57245



A-A

Air Inlet Seal Inspection
Figure 601 (Sheet 2)

CUSTOMER FLEET EFFECTIVITY

ALL

SOURCE

MRB

SIGMA SEAL

49-030-00-00

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DATE		TAIL NUMBER		STATION		AIRLINE CARD NO.		BOEING CARD NO. 49-040-00-00		
SKILL ENGIN	WORK AREA APU COMPARTMENT	RELATED TASK		VERSION 1.1	THRESHOLD APU CNG	REPEAT		PHASE		
TASK INSPECTION		TITLE APU INSULATION PANELS				APPLICABILITY AIRPLANE ENGINE ALL ALL				
ZONES 315 316			ACCESS 315A							
<p>SYSTEMS</p> <p>MPD ITEM: 49-040-00 PERFORM A DETAILED INSPECTION OF THE APU INSULATION PANELS. (AFTER APU REMOVAL).</p> <p>A. Consumable Materials (1) G01288 Tape, Carpet - BMS5-133 (2) G02305 Tape, Insulation Blanket - BMS5-149</p> <p>B. References (1) AMM TASK 49-11-00-000-801 p401, APU Power Plant Removal (2) AMM TASK 49-11-00-400-801 p401, APU Power Plant Installation (3) AMM TASK 49-17-11-000-801 p401, Insulation Panel Removal (4) AMM TASK 49-17-11-300-801 p801, Repair of the Insulation Panel (5) AMM TASK 49-17-11-400-801 p401, Insulation Panel Installation</p>								MECH	INSP	
CUSTOMER FLEET EFFECTIVITY ALL				SOURCE MRB	APU INSULATION PANELS					
				49-040-00-00					PAGE 1 of 6 Oct 10/01	

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-040-00-00		
<p>TASK 49-17-11-200-801</p> <p>1. <u>Insulation Panel Inspection</u></p> <p>A. Prepare for the Inspection</p> <ul style="list-style-type: none"> (1) Make sure the APU master switch on the P5 forward overhead panel is OFF and install a DO-NOT-OPERATE tag. (2) Open these circuit breakers and attach DO-NOT-CLOSE tags: <ul style="list-style-type: none"> (a) Circuit Breaker Panel, P6-4: <ul style="list-style-type: none"> 1) 6A14 AUX POWER UNIT CONT (b) Circuit Breaker Panel, P6-2: <ul style="list-style-type: none"> 1) 6B19 APU FIRE SW POWER (3) Open the APU Cowl Door, 315A: <ul style="list-style-type: none"> (a) Open the three latches. (b) Open the APU Cowl Door, 315A. (c) Remove the retainer pin from the rod end of the forward hold-open rod on the APU Cowl Door, 315A. (d) Remove the retainer pin from the spring clip on the aft hold-open rod. (e) Disconnect the two hold-open rods from the two spring clips. (f) Connect the two rod ends of the two hold-open rods to the two brackets in the APU compartment. (g) Install the two retainer pins in the two rod ends. (4) If it is necessary to remove the APU to do a full inspection of the seven insulation panels, do this task: APU Power Plant Removal (AMM TASK 49-11-00-000-801 p401). <p><u>NOTE:</u> It is necessary to remove the APU to inspect all the surfaces of the top insulation panel.</p> <p><u>NOTE:</u> It is not necessary to remove the APU if you do a general visual inspection of the seven insulation panels.</p> <p>B. Procedure</p> <ul style="list-style-type: none"> (1) BEJ 001-007; <p>Do these steps to inspect the insulation panels for fluid contamination and structural damage:</p> <ul style="list-style-type: none"> (a) Examine the insulation panels for signs of fluid contamination to the core insulation material. (b) Examine the surface of the insulation panels for missing weld stitches, missing metal sheets, ruptured seams or structural metal deterioration. (c) Examine the insulation panels for holes that have gone through the inner and outer metal sheets. 					MECH	INSP
CUSTOMER FLEET EFFECTIVITY ALL			SOURCE MRB	APU INSULATION PANELS 49-040-00-00 PAGE 2 of 6 Oct 10/02		

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-040-00-00	
<p>(d) If you find any of the above damage, replace the insulation panel(s). These are the tasks: Insulation Panel Removal (AMM TASK 49-17-11-000-801 p401), Insulation Panel Installation (AMM TASK 49-17-11-400-801 p401). 1) After the insulation panel(s) are removed, examine the foam and support insulation for contamination and damage.</p> <p><u>NOTE:</u> You can find the foam and support insulation behind the insulation panel(s).</p> <p>2) If you find contamination or damage, remove the foam and/or support insulation. 3) If it is necessary, examine the structure behind the support insulation for contamination and damage. a) Clean and repair the problems that you find. 4) Replace the foam and/or support insulation before you install the insulation panel(s).</p> <p><u>NOTE:</u> Use tape, BMS5-133 or tape, BMS5-149 to install the support insulation to the bulkhead.</p> <p>(2) BEJ 008-999; Do these steps to inspect the insulation panels for fluid contamination and structural damage: (a) Examine the insulation panels for signs of fluid contamination to the core insulation material. (b) Examine the surface of the insulation panels for missing weld stitches, missing metal sheets, ruptured seams or structural metal deterioration. (c) Examine the insulation panels for holes that have gone through the inner and outer metal sheets. (d) If you find any of the above damage, replace the insulation panel(s). These are the tasks: Insulation Panel Removal (AMM TASK 49-17-11-000-801 p401), Insulation Panel Installation (AMM TASK 49-17-11-400-801 p401). 1) After the insulation panel(s) are removed, do these steps: a) Examine the structure behind the insulation panel(s) for contamination and damage. b) Examine the air inlet scoop for blockage of unwanted materials and damage that can cause a decrease in air flow.</p> <p><u>NOTE:</u> You can find the air inlet scoop behind the forward insulation panel.</p> <p>c) If you find blockage of unwanted materials, remove the blockage.</p>				MECH	INSP
CUSTOMER FLEET EFFECTIVITY ALL		SOURCE MRB	APU INSULATION PANELS 49-040-00-00 PAGE 3 of 6 Feb 10/02		

BEJ



DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 49-040-00-00		
<p>d) If you find contamination or damage, clean and repair the problems that you find.</p> <p>(3) BEJ 001-007; Do these steps to inspect the insulation panels for other structural damage limits:</p> <p>(a) Examine the outer metal sheet on the insulation panel for holes and tears.</p> <ol style="list-style-type: none"> 1) Holes less than 0.25 inch (6.4 mm) in diameter. 2) Tears less than 4 inches (102 mm) in length. <p>(b) Examine the damaged areas of the insulation panels.</p> <ol style="list-style-type: none"> 1) If you find more than 0.5 inch (13 mm) of the outer metal sheet around the damaged area(s), you can repair the insulation panel. <p><u>NOTE:</u> This limit includes the distance from the damaged area to a grommet, sharp bend, edge of the metal sheet or attaching parts.</p> <p>(c) If you find the above damage and the damage is in the limits, do this task: Repair of the Insulation Panel (AMM TASK 49-17-11-300-801 p801).</p> <p>(d) If you find the above damage and the damage is more than the limits, replace the insulation panel(s). These are the tasks:</p> <p>Insulation Panel Removal (AMM TASK 49-17-11-000-801 p401), Insulation Panel Installation (AMM TASK 49-17-11-400-801 p401).</p> <ol style="list-style-type: none"> 1) After the insulation panel(s) are removed, examine the foam and support insulation for contamination and damage. <p><u>NOTE:</u> You can find the foam and support insulation behind the insulation panel(s).</p> <ol style="list-style-type: none"> 2) If you find contamination or damage, remove the foam and/or support insulation. 3) If it is necessary, examine the structure behind the support insulation for contamination and damage. <ol style="list-style-type: none"> a) Clean and repair the problems that you find. 4) Replace the foam and/or support insulation before you install the insulation panel(s). <p><u>NOTE:</u> Use tape, BMS5-133 or tape, BMS5-149 to install the support insulation to the bulkhead.</p>					MECH	INSP
CUSTOMER FLEET EFFECTIVITY ALL			SOURCE MRB	APU INSULATION PANELS 49-040-00-00 PAGE 4 of 6 Oct 10/01		

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